ME, ECE Capstone Design Programs

Team 21 – SAE Aero Design Micro Class

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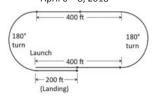
OBJECTIVE STATEMENT

The goal of this project is to design a competition winning RC aircraft under the constraints dictated by the 2018 SAE Aero Design Rules and Regulations. The team is to minimize aircraft weight, maximize payload, and reduce assembly time through utilization of the design process.

BACKGROUND INFORMATION

Competition Details

- Van Nuys, CA
- April 6 8, 2018



Competition Imposed Constraints

- Payload: 2" SCH 40 PVC Pipe
- Hand launched
- Electric Motor/Lithium Polymer Battery
- Aircraft, payload, and tools stored in container
- · Flight circuit (left)

Category	Score	Place
Overall Score	92.3891	4 th
Payload Fraction	0.69	3 rd
Flight Score	29.5852	5 th
Design Report	31.8640	6 th
Technical Presentation	37.9400	9 th

COMPETITION RESULTS

ASSEMBLED DESIGN





BUDGET



Jack Rettig



TESTING

Test	Expected Result	Measured Result
Static Thrust Test	> 140 grams	430 grams
Wing Break Test	1.48 lbs.	7.11 lbs.
Wing Deflection Test	1.7"	1.5"
Battery Life Test	> 30%	58% Remaining
Servo Test	4.16 ozin.	10.86 oz./ 60°

SAFETY

- Red arming plug to arm and disarm system safely
- Programmed throttle failsafe to automatically shut system down if signal is lost
- Lithium Polymer battery charging bag to protect from battery fires
- Painted propeller tips
- Warming up and wearing a hard hat during launch

MATERIALS AND MANUFACTURING

Component	Material/Manufacturing	
Wing/Tail Ribs	Balsa Wood – laser cut from balsa wood sheets	
Wing Joints	Carbon Fiber – rods purchased and cut with bandsaw	
Wing Spars	Balsa Dowels – ordered and cut to length with bandsaw	
Wing Bracket	Balsa Wood – pieces laser cut individually; glued together; sanded to shape	
Central Rods	Carbon Fiber Tubing – premade (ordered); cut to length with bandsaw	
Wing Bracket/Rib Reinforcement	Carbon Fiber Tubing – premade (ordered); cut to length with bandsaw	
Wing/Tail Skin	Coverite Microlite Film – cut to shape; ironed onto balsa wood ribs	
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Advisers: Jack Hawkins, Mark Rabalais, Drew Rousseau

ENGINEERING SPECIFICATIONS

Measurable Specification	Value	Description
Max Total Weight	10 lbs.	Weight of aircraft, payload, container, and assembly tools
Container Dimensions	13.875" x 12.125" x 3.625" t = 0.125"	Dimensions of cardboard carrying container
Container Weight	1.5 lbs.	Weight of empty cardboard carrying case
Payload Dimensions	OD = 2.375", ID = 2"	Outer/Inner diameter of PVC pipe
Payload Specific Gravity	0.68 lbm./ft.	Density of PVC pipe per linear foot
Battery Size	2-cell 325 mAh	Optimized battery size
Assembly Time	1:45 min	Aircraft assembly time
Flight Time	60 s	Maximum allowed time of flight
Project Start Resea	rch Configuration De	uselage sign and Layout Propulsion System
Aug. 26, 2017 Sep. 2	017 Sep. 2017 Se	p. 2017 Oct. 2017

Sponsors: Dr. K. Gonthier, LaSPACE, Mr. Jack Rettig, LSU MIE/EE Dept.